

# Saint Martin's

# 2017

# PCI BIG BEAM COMPETITION



### SAINT MARTIN'S UNIVERSITY: KRAKEN AGAIN

### **Student Members**

Cameron Reece, William Miller, Paul Rumbles, Jarad Roschi, Joel Rogers, Clarinda Marion, David Rowland

**PCI Producer Member** Concrete Technology Corporation

**Faculty Advisor** Jill Walsh, PhD, PE

### PCI BIG BEAM COMPETITION 2016-17

#### June 12, 2017

Date		May 5, 2017
Saint Martin's University Student Team (school name)	L Team Number	Date of Casting
Basic Information       28         1. Age of beam at testing (days)       28         2. Compressive cylinder tests*	inad)/calculah b. Absolutie value deflection//cal c. Absolutie value inad//calculah Total of three a <sup>1</sup> Measured cracking ina inad/deflection curve. F	pplied load (kip) 34.88 3 load (kip) <sup>4</sup> 24.44 168.54 1690 deflection (in.) 5.44

### Test summary forms must be included with the final report, due June 16, 2017.

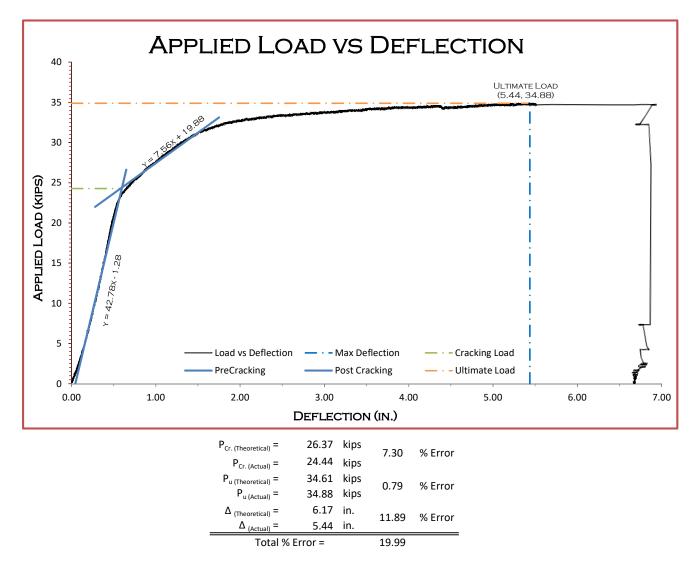
Sponsored by:







RESULTS





# **PCI BIG BEAM COMPETITION 2016-17**

### CERTIFICATION

# As a representative of (name of PCI Producer Member or sponsoring organization)

SAINT MARTIN'S UNIVERSITY Sponsoring (name of school and team number) TEAM

### I certify that:

- . The beam submitted by this team was fabricated and tested within the contest period.
- The calculations of predicted cracking load, maximum load, and deflection were done prior to testing of the beam.

1

- The students were chiefly responsible for the design.
- The students participated in the fabrication to the extent that was prudent and safe. .

Mare

MAUE, P.E.

The submitted test results are, to the best of my knowledge, correct, and the video submitted is of the . actual test.

### **Certified by:**

in Signature

AUSTIN Name (please print)

6/12/17 Date

### THIS CERTIFICATION MUST BE PART OF THE FINAL REPORT

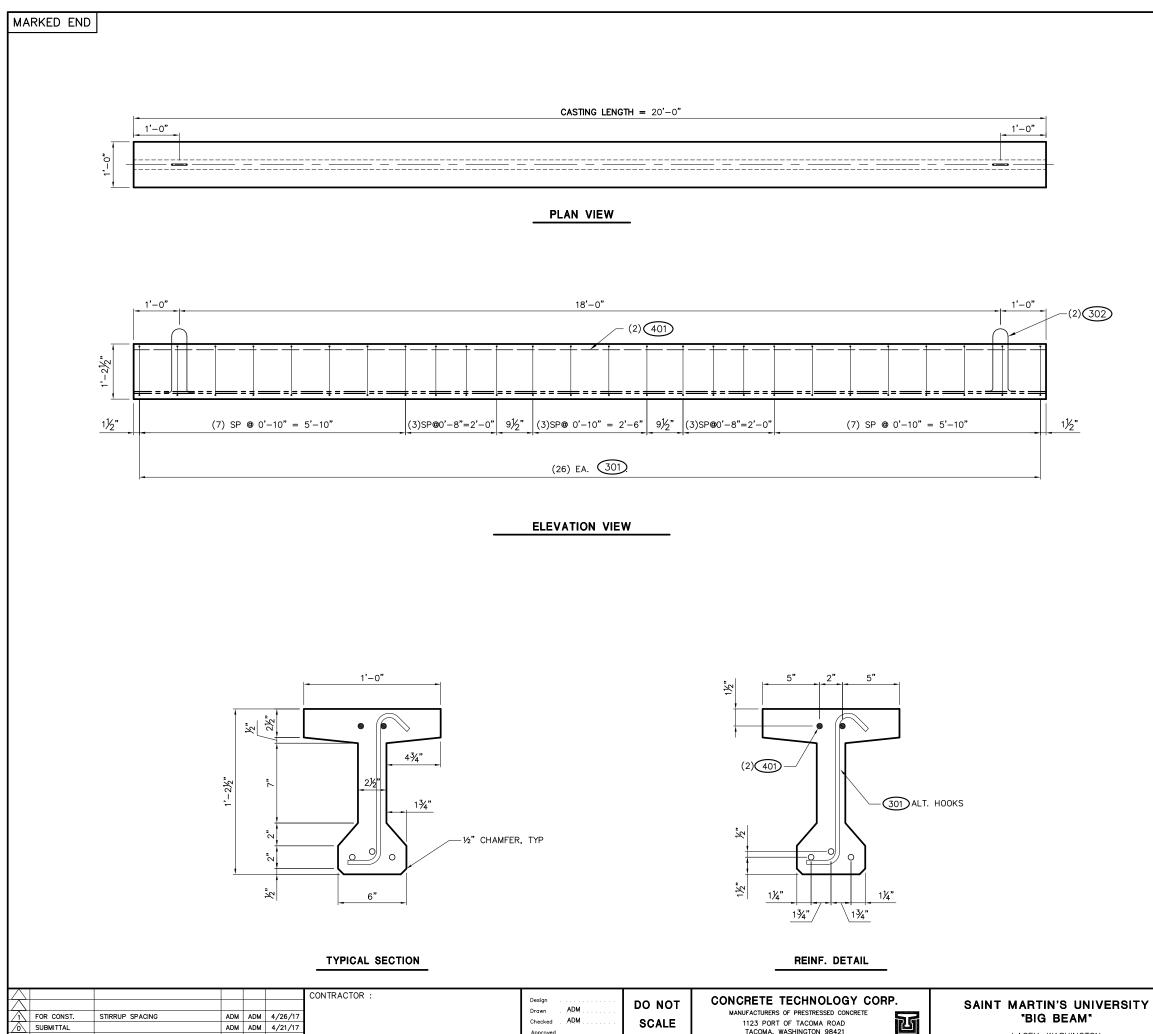
Sponsored by:

AUTODESK 'S SOLUTION ASSOCIATE FOR PRECAST



**BUILDING TRUST** 





No. Status

Revision

Design Drawn Checked ADM MANUFACTURERS OF PRESTRESSED CONCRETE 1123 PORT OF TACOMA ROAD TACOMA, WASHINGTON 98421 ADM ADM 4/26/17 Ŀ "BIG BEAM" ADM SCALE 
 ADM
 ADM
 4/21/17

 By
 App'd
 Date
 Approved LACEY, WASHINGTON

	REINFORCEMENT SCHEDULE										
GRADE	BAR MARK NUMBER	QUANTITY	STRAIGHT	NO. BENDS	CUTTING LENGTH	REBAR TYPE	BAR MAR	SIZ K N	E UMBER	<u>PIN</u> #3	$\frac{RUP/TIE}{\phi (UNO)} = 1\frac{1}{2}^{n}$ $= 2^{n}$
60	301	28		2	1'-6"					#5	= 21/2"
H	302 401	2	X	3	3'-5½" 20'-4"	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~					= 41/2" = 51/4"
H		2	ŕ	$\vdash$	20-4	Ś Ś	ŧ				= 594 $= 6^{"}$
Ħ	$\sim$								L		
$\vdash$	$\leq$					-1%"					
H	$\succ \prec$	]	┝								
H	$\succ \prec$	1	┢	┢							
H	$\geq$		t								
$\square$	$\square$					27⁄8"					
$\vdash$	$ \searrow $					(301)					
$\mathbb{H}$	$\succ \prec$		┢								
H	$\geq$										
$\square$	$\sim$										
$\mathbb{H}$	$\succ \prec$		┝	$\vdash$			ł			4"ø PIN	
H	$\geq$		┢								
t	$\sim$					- 434"					
Γ							1		Jl		11⁄2"ø PIN
1							<u>!</u>			_	11.1
								2"		2"	
								(	302	)	
									$\sim$		
1											
1											
	GENER	AL NO	TE	S							
-	ONCRETE:					7,000 P.S.I.					
ľ						. <u>13,000</u> P.S.I.					
<u> </u>	RESTRESS				, 7 WIRE U TION STRAN						
1		2011 NE				-					
1	STRAIGH	T: (3) STF	1AS	ND:	S JACKED T	0 93 KIPS (31.0 K/STR.)					
FI	NISHES:	TOP	S	TEF	L TROWEL						
"		SIDES	FC	DRM	I FINISH						
1		SOFFIT ENDS									
					PING BUNK						
_					FROM EAC						
<u>s</u>	HIPPING B	UNKING: 1	-	U″	FROM EAC	I LNU					
F	INSER	Г & AS	SE	EN	IBLY SC	HEDULE					
⊢					-						
$\vdash$	+						+	$\vdash$			
$\vdash$	$\rightarrow +$						-	$\vdash$			
⊢	$\searrow$						-				
L											
L											
L											
F							1	Ħ	PER		TOTAL
N	IARK				DESCR	PTION	PRIME	GALV.	PIECE	E	
⊢		6 . 11		_		21-11					TIES
	for which	e of these ch they we	pl ere	an pr	s and spec repared. A	fications shall be restricted to ny reproduction or distribution is	ine sex	orig	inal pu ssly lim	irpos ited	to
	such us part, w	se. Any o ithout the	th wr	er itt	reproduction en consent	ny reproduction or distribution is n, reuse or disclosure by any m of CTC is prohibited. These dr	etha awir	od, ngs	in whol and	e or	in
⊢	specific	ations con	τα	n	proprietary	information and title remains in	UI	U.			
L		PRO	DI	JC	TION D				APRIL 20		
B	EAMO <sup>.</sup>	THR				Sh	. No.	 , F	of 3M—2	· · · · ·	
<b> </b> №	IARK:_	BM2		Q	<b>TY.</b> <u>1</u>	_ WT. <u>1.55</u> KIPS [ 🦷	9. No.	1	17X08	ЗA	· · · · · · · · ·

### TABLE OF CONTENTS

	_	
Introduction	7	
Acknowledgements	8	
Assumptions	8	
Materials	8	
Concrete		8
Prestressing Strands		
Rebar		
Design Process		
Design Concept		9
Flexural Design		10
Shear Design		10
Beam Fabrication		
Reinforcement Construction		11
Strand Prestressing and Beam Casting		12
Testing		
Results	14	
Lessons Learned	16	
Appendices	18	
Appendix A		19
Drawings, Formwork, and Line Layout		19
Appendix B		23
Weight and Cost		23
Appendix C		25
Structural Design and Analysis Calculations		25
Appendix D		35
Concrete Mix Design		35
Appendix E		37
Strand Specifications		37
Contact Information	40	

## **INTRODUCTION**

In 2017, Saint Martin's University (SMU) entered its second Big Beam Competition under the supervision of faculty advisor Jill Walsh, PhD, PE. The tasks to achieve in the competition were to design, construct, and test a prestressed concrete beam according to the criteria laid out by Precast/Prestressed Concrete Institute (PCI) while meeting the American Society for Testing and Materials (ASTM) as well as the American Concrete Institute (ACI) standards.

The team's goal was to design a simple cross section that would behave as expected. The tested design was an I-shaped beam of constant depth and cross-section with normal weight, high-strength concrete, three prestressing strands, two longitudinal reinforcing steel bars, and alternating Z-shaped stirrups for shear reinforcement. Most of the design was done with a spreadsheet created by the 2016 Saint Martin's University' Big Beam team. The spreadsheet simultaneously calculates stresses in the concrete, strands, and rebar, employs a macro to find the equilibrium of internal forces, and generates the moment curvature of a cross section. The spreadsheet required a few coding adjustments as well as the inclusion of an additional calculations sheet to calculate release stresses. The team also added an automated section properties sheet to assist with efficiency in the design process. A detailed description of the beam is in the DESIGN PROCESS section. A comparison of predictions and actual results is shown below in TABLE 1.

### TABLE 1. PREDICTIONS VERSUS RESULTS

	Prediction	Results
Ultimate Load (kips)	34.61	34.88
Deflection at Ultimate Load (in)	6.17	5.44
Cracking Load (kips)	26.37	24.44

While the ultimate load prediction was within 0.8% and the cracking load was within 7.3%, the deflection prediction was 11.9% lower than predicted. A detailed discussion about the possible causes of inaccuracies is in the RESULTS section.



## ACKNOWLEDGEMENTS

Saint Martin's University' PCI Big Beam team is very thankful for the opportunity to participate in this unique competition. The team would not have been able to successfully design, build, and test the beam without the help of its advisor and sponsors.

Thank you to Dave Chapman, PE and Concrete Technology Corporation for choosing to sponsor Saint Martin's University's second competition entry. Additionally, thank you to the employees at CTC who donated their time and effort to help construct the beam.

The team is especially grateful to Austin Maue, PE for providing an incredible amount of advice throughout the competition.

Thank you Saint Martin's University for the support received during the course of this competition, and Jill Walsh, PhD, PE, this year's faculty advisor.

Finally, thank you to John Stanton, PhD, and the University of Washington for hosting the team and allowing the use such an impressive testing facility.

## ASSUMPTIONS

- Strands are fully bonded with concrete. Strain changes in the steel and concrete are the same at strand release.
- Actual stress-strain relationships and materials are very similar to the constitutive models used.
- Strains are distributed linearly over the depth of the cross section.
- The ultimate moment is based on the strain which causes either concrete crushing or strand fracture.
- Members fail in flexure.

## MATERIALS

### Concrete

The decision to enter the competition was made a little later than what would have been convenient which limited the amount of options available for concrete mix. There were three options of concrete mix; high-strength, normal-weight, or light-weight. The high strength concrete is more expensive than normal weight, but extra cost is mitigated by using less material (requiring less cross-sectional area). Light-weight concrete provides less dead weight but is also quite expensive. After debating the benefits of each type of available concrete mixes, the team chose a normal-weight, high-strength concrete mix. In future years, the design process should begin with enough time to make the proper arrangements to use a mix design specified by the team.

The chosen mix for the beam this year is used regularly for projects at Concrete Technology Corporation (CTC). The mix had a 0.27 water/cement ratio, a slump of 7.00 inches, an air content of 1.4%, and a unit weight of 152.5 pcf. TABLE 2 shows a summary of the mix; additional details can be found in APPENDIX D. The mix performed well for the teams requirements, the original design for the beam was to have an initial compressive strength of 7,000 psi and the actual mix surpassed that by more than enough.

<b>Cementitious Materials</b>	Aggregates	Admixtures	Concrete Strength (psi)
750 lb Type III Cement	1,993 lb Course	1.9 lb WDRA 64	f'ci: 10,650
	1,264 lb Fine	4.2 lb ADVA 575	f'c: 13,505
			Tensile Strength: 1,670

### TABLE 2. CONCRETE MIX FOR ONE CUBIC YARD OF CONCRETE

### **Prestressing Strands**

The Prestressing strands used in the beam were low relaxation <sup>1</sup>/<sub>2</sub>" diameter ASTM A415 grade 270 strands. Sumiden Wire Products Corporation provided strand certifications that produced a yield point of 40.25 kips and a modulus of elasticity of 28,900 ksi. The strands were a constant depth for the total beam length. Detailed strand properties are in APPENDIX E.

### Rebar

Two #4 bars, ASTM A615 grade 60 rebar, longitudinally continuous, were used in the top flange. The longitudinal bars were used to hold the shear reinforcement as well as to increase the tension capacity in the top flange. The design also included #3 ASTM A615 grade 60, Z-shaped stirrups for shear reinforcement spaced at  $8^{\circ} - 10^{\circ}$ .

### **DESIGN PROCESS**

### Design Concept

The goal this year was accuracy and maximizing deflection. Using an in house developed moment curvature analysis to predict the behavior of the beam, the team decided it would be best to optimize the accuracy of the theoretical calculations for future competitions before introducing other innovative factors such as; de-bonding, varied cross-section, harped strands, etcetera... The team considered two common cross-sections, T-shaped and I-shaped. After debating the pros and cons of each, the first design was a T-beam. The team performed the moment-curvature analysis on multiple variations of T-beams to finalize cross-section.

The next step was to analyze stresses directly following strand release. During the release-stress analysis, the assumed initial concrete compressive strength (f'ci of 7,000 psi) combined with the

provided area of concrete below the neutral axis was not enough to resist the compression force applied by the prestressing strands. This is what caused the addition of the bottom flange. After adding a bottom flange, an analysis was performed on many iterations of an I-beam before deciding on the final crosssection shown below in FIGURE 1.

When designing a beam in a real-world scenario, it is important that the member deflects during the yielding phase of failure. Without deflection, hairline cracks may remain unnoticed by the general population causing a surprise failure that may cause serious injury or death. So the final decision of the cross section was made based on maximizing the predicted deflection.

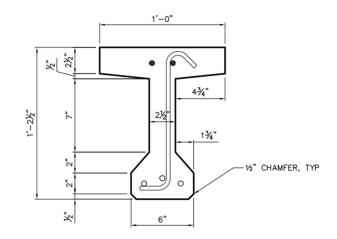


FIGURE 1. TYPICAL BEAM CROSS-SECTION

### Flexural Design

Strand sizes were readily available in 0.5 inch diameter and 0.6 inch diameter. The smaller strands provided a larger variety of choices in the strand layout, which is why 0.5 in. dia. strands were chosen. Using an assumed strand stress at failure of 270 ksi and through manipulation of the strand locations, the final configuration was established. The ultimate decision was made based on a service load of 20.0 kips and a maximum load range of 32.0 - 39.0 kips. The team chose to aim for values in a few kips above the threshold for cracking and in the middle of the ultimate range to allow for some discrepancies between prediction and actual values while avoiding penalties. Complete design and fabrication drawings are included in APPENDIX A.

The two #4 longitudinal bars in the top flange provide sufficient tensile capacity during and after the release of the prestressing strands until the service load is applied.

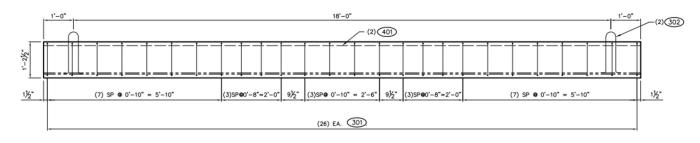
### Shear Design

Flexure failure has many signs of distress, such as large deflections and cracking, whereas shear failure often occurs with little warning and is catastrophic. For this reason the design is conservative in shear. The team designed shear reinforcement using the maximum load allowed by the competition, 39 kips.

While this would add additional steel, and therefore cost, the guarantee of shear strength was deemed worth it.

The shear strength of the concrete was calculated based on the ACI 318-11 code, therefore the shear design was determined from the lesser of V c i, V c w and  $(2)\sqrt{f'c} (b w)(d)$ . Understanding that V c i and V c w varies along the length of the beam as the prestressing force develops and loading changes, these values were looked at, and many other points along the length of the beam. Key points such as h/2, l t, l d and mid-span, as well as ranges in between these points, were analyzed to produce a complete picture of the beams need for shear design. The shear reinforcement comprised alternating Z-shaped stirrups, placed at 10" for the end 5'-10" feet, 8" for the next 2 feet, one space of 9½", and 10" spacing for the middle 2'-6" of beam, as displayed in FIGURE 2 below.

### FIGURE 2. REINFORCEMENT ELEVATION



## **BEAM FABRICATION**

Reinforcement Construction

The formwork was designed and constructed by CTC after the submittal of the final design and May 5<sup>th</sup>,

2017 SMU's team was invited to CTC to assemble the rebar cage. Prior to the team arriving, CTC ran the three 1/2" diameter prestressing strands through the wooden end plates and stressed them. To begin construction, the team measured and marked the designed stirrup



spacing along the pre-stressing pallet. To assemble the rebar cage, the team used zip ties to position the alternating Zshaped stirrups. At both ends and at the mid-span, the stirrups were tied together in alternating directions and then tied to the two longitudinal #4 bars. The two #4 bars were cut a little too long to allow them to pass through the end-piece formwork; the purpose of this was to help hold the longitudinal rebar in place while the stirrups were tied. Once the stirrups were set, the side forms of the beam were positioned and all seams of the formwork were sealed to prepare for the concrete pour.

### Strand Prestressing and Beam Casting

The beam casting was completed by CTC. After the concrete



was poured, the crew finished the top of the beam with a steel trowel. A thermal monitoring device, was inserted, and left in the wet concrete that was



linked to a computer that monitored the temperature controlled test cylinders to ensure the cylinders cured at the same temperature as the beam. The beam was then covered and left to set for 92 hours before strand release and removal from its frame. The beam cured under damp burlap and a plastic covering to avoid uneven moisture loss. CTC tested two cylinders on the same day of release to record the initial compressive strength and. The beam was 28 days old at testing.

## TESTING

One day prior to testing, CTC tested two additional cylinders and recorded the final compressive strength of the beam. See TABLE 2 for recoded compressive strength. Testing occurred on June 2, 2017, at the University of Washington's More Hall Structures Lab. Prior to SMU arriving, the lab technicians assembled the following test rigging beneath the Baldwin machine.



The locations of the supports, the mid-span, and the two point loads were marked on the beam. A steel bar was hot glued into place at the centerline of the mid-span mark beneath the beam and strings tied to the ends of the bar were attached to a potentiometer underneath the beam to measure deflection. As a back-up, in case the computer recording the data malfunctioned, a meter stick was placed vertically beside one of the two point loads and multiple video cameras were placed around the beam during testing. Along with the video cameras, tasks were assigned to each individual: data acquisition, videography, watching for cracking, and recording the time and load at which they occurred.

The electronic testing instruments were calibrated before the test began. The plan for the test was to load the beam to the service load of 20 kips and check for cracking, if the beam passed the inspection of CTC representative, Austin Maue, PE, the beam would be completely unloaded. The load application would begin again and the beam would be tested until failure. If cracks were noticed, the load application would continue from where it stopped without being unloaded because if the beam has cracked it will no

longer behave elastically, and the actual ultimate load and maximum deflection would not be accurate. Once the test began, the Baldwin machine was set to approach 20 kips in a timeframe of 3 minutes. The beam passed the crack inspection and the load was removed; the beam was then loaded until failure. As the beam approached failure, it experienced flexural cracking in the central 3 - 4 feet of the beam. The beam eventually failed in compression in the top flange at a load of 34.88 kips and a corresponding deflection of 5.44 inches.



## RESULTS

APPLIED LOAD VS DEFLECTION 40 ULTIMATE LOAD (5.44, 34.88) 35 30 APPLJED LOAD (KIPS) 10 5 Load vs Deflection Max Deflection PreCracking Cracking Load Post Cracking Ultimate Load 0 7.00 0.00 1.00 2.00 3.00 4.00 5.00 6.00 DEFLECTION (IN.)

FIGURE 3. LOAD- DEFLECTION CURVE

After testing, the data was collected from the computer and used to plot the load-deflection curve shown

### in FIGURE 3. The portion of the curve preceding the cracking load is quite linear making a linear approximation reasonably simple. However, following the cracking load, the graph is a nonlinear progression which made the linear approximation of the post cracking slope somewhat difficult compared to the pre-cracking slope. After analyzing the data, the team recorded the actual numbers shown in TABLE 3.

	Prediction	Results	<b>Error Analysis</b>
Ultimate Load (kips)	34.61	34.88	0.79%
Deflection at Ultimate Load (in)	6.17	5.44	11.89%
Cracking Load (kips)	26.37	24.44	7.30%
Total			19.99%

### TABLE 3. ERROR ANALYSIS

As apparent in TABLE 3, the predictions were rather close to the results. In 2016 the cracking load for "The Kraken" was calculated with the same moment curvature analysis spreadsheet with only 0.34% error and the ultimate load had an error of 6.65%. The original assumption was that the error in the predictions this year shifted from the ultimate load to the cracking load, but after careful consideration the team decided that the error in the cracking load was due to the system used to cure the test cylinders compared to the curing of the beam itself. The cylinders were cured in a lime bath at the CTC plant and the beam cured in the storage yard outside. This is what caused the cylinder's modulus of rupture (MOR) to be 1,670 psi instead of the actual MOR of approximately 1,300 psi. When the team used 1,300 psi as the MOR, the post-testing predictions were much more accurate (shown below in TABLE 4). As for the high deflection prediction, an error was made in assuming that the physically measured camber needed to be added to the calculation of deflection caused by the ultimate load. The reason for adding this has to do with the integration process used for determining the deflection of the beam. The

team integrated the curvature of the beam twice to calculate the deflection and when integrating twice, there are two constants of integration. It was rational that one constant is found by performing the same procedure of calculating deflection considering only the self-weight of the beam, a zero-load case, and the second constant was the physically measured camber of the beam. After testing the beam it is apparent that the physically measured camber is already accounted for in the deflection calculation of the zero-load case. The total camber was measured to be about 0.875 inches; when this is subtracted from the prediction and after adjusting the MOR, the predicted deflection is rather precise (displayed in TABLE 4 and FIGURE 4).

	Prediction	Results	<b>Error Analysis</b>
Ultimate Load (kips)	34.61	34.88	0.79%
Deflection at Ultimate Load (in)	5.33	5.44	0.06%
<b>Cracking Load (kips)</b>	24.46	24.44	2.00%
Total			2.85%

TABLE 4. ERROR ANALYSIS WITH AN MOR OF 1300 PSI

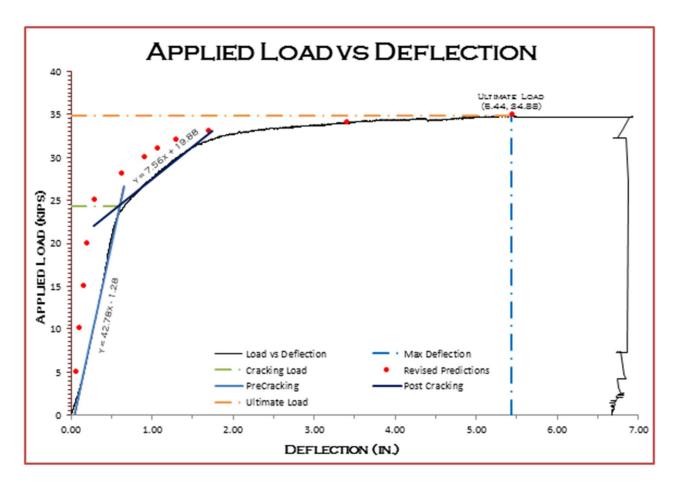


FIGURE 4. RE-EVALUATED PREDICTIONS GRAPH

## LESSONS LEARNED

The team learned many lessons throughout the design, construction and during the data analysis. The first lessons were learned in the design process. During the initial cross-section comparisons, the differences between a T-beam and an I-beam were not apparent. It was not until the addition of release stress analysis to the Excel spreadsheet that the team noticed the significance of the bottom flange. However, in the real-world pre-stressed I beams, the bottom flange is typically only increased to allow for the addition of more prestressing strands which adds more moment capacity and lengthens the maximum span of a cross-section.

During construction, the team did not encounter any major difficulties in assembling the reinforcement cage. The only challenge was that most of the bent Z-shaped stirrups were a fraction too long. This caused the longitudinal bars to rise vertically out of the minimum clear cover in the top flange. The solution to this problem was to tilt each of the stirrups enough to bring the longitudinal bars to the desired height. The tilting was administered at an angle to intersect shear cracks. This is when the team realized how a singular minor discrepancy of the specified design could cause complications in fabrication which revealed the reason for simple design specifications. The team also became more familiar with the system used to jack the prestressing strands, how stirrups were placed, and overall how the CTC plant operated.

Through analyzing the results, the team gained a better understanding of how a moment-curvature analysis works and how it is used to predict ultimate deflection. As mentioned in the RESULTS section the team made an error in determining the constants of integration and by analyzing the predictions versus the actual data, the error was discovered. This should not have been an issue this year due to the fact that the same process for the deflection calculation was used in the 2016 PCI Big Beam Competition entry and any error should have been found then. However in 2016, the deflection calculation was 32.37% below the actual deflection and not above as it was this year, which is why the error in the integration constant was not apparent. This year the team decided that the reason for the error in 2016 was that the setup of the single point-load was not a true point-load and the load distribution to the top flange produced confinement in the compression region of the concrete, where the failure was designed to occur. This resulted in a 6.7% larger ultimate load which caused the deflection to be larger than predicted. Confinement of concrete was not an issue this year because the load was applied as two point-loads offset from the mid-span and therefore not directly applied to the designed failure region.

Finally, it became evident that communication is paramount throughout the entire process; from brainstorming design ideas and developing clear drawings to scheduling meetings and coordinating with CTC and UW for the build and test dates.

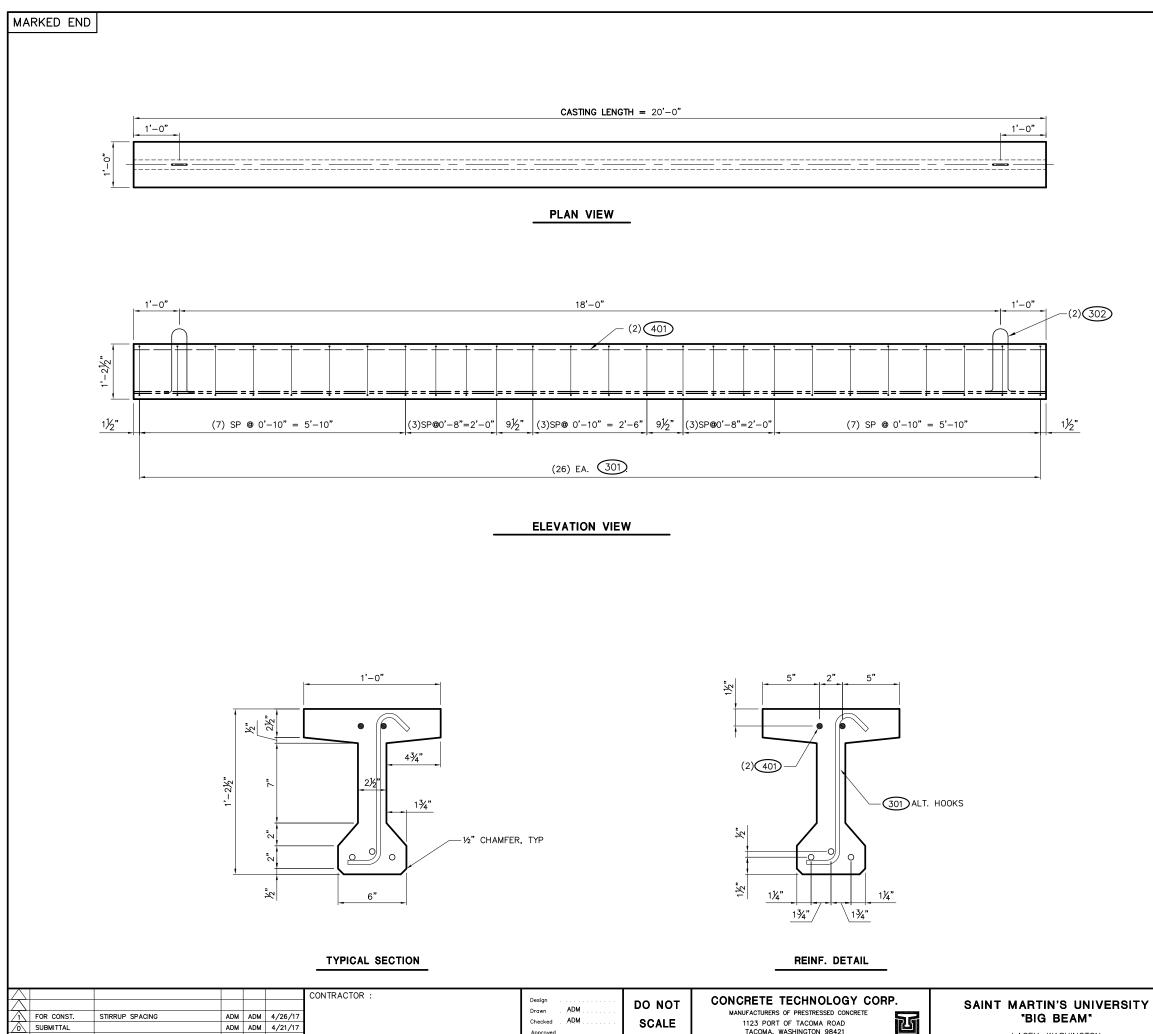
It is a rare and treasured opportunity to be able to design, build, and test a prestressed concrete product to the point of failure. The team is honored and thankful for the opportunity to participate in the competition. The team learned many lessons and gained experience that will better prepare them as they move forward in their future engineering careers.



# APPENDICES

# Appendix A

# Drawings, Formwork, and Line Layout



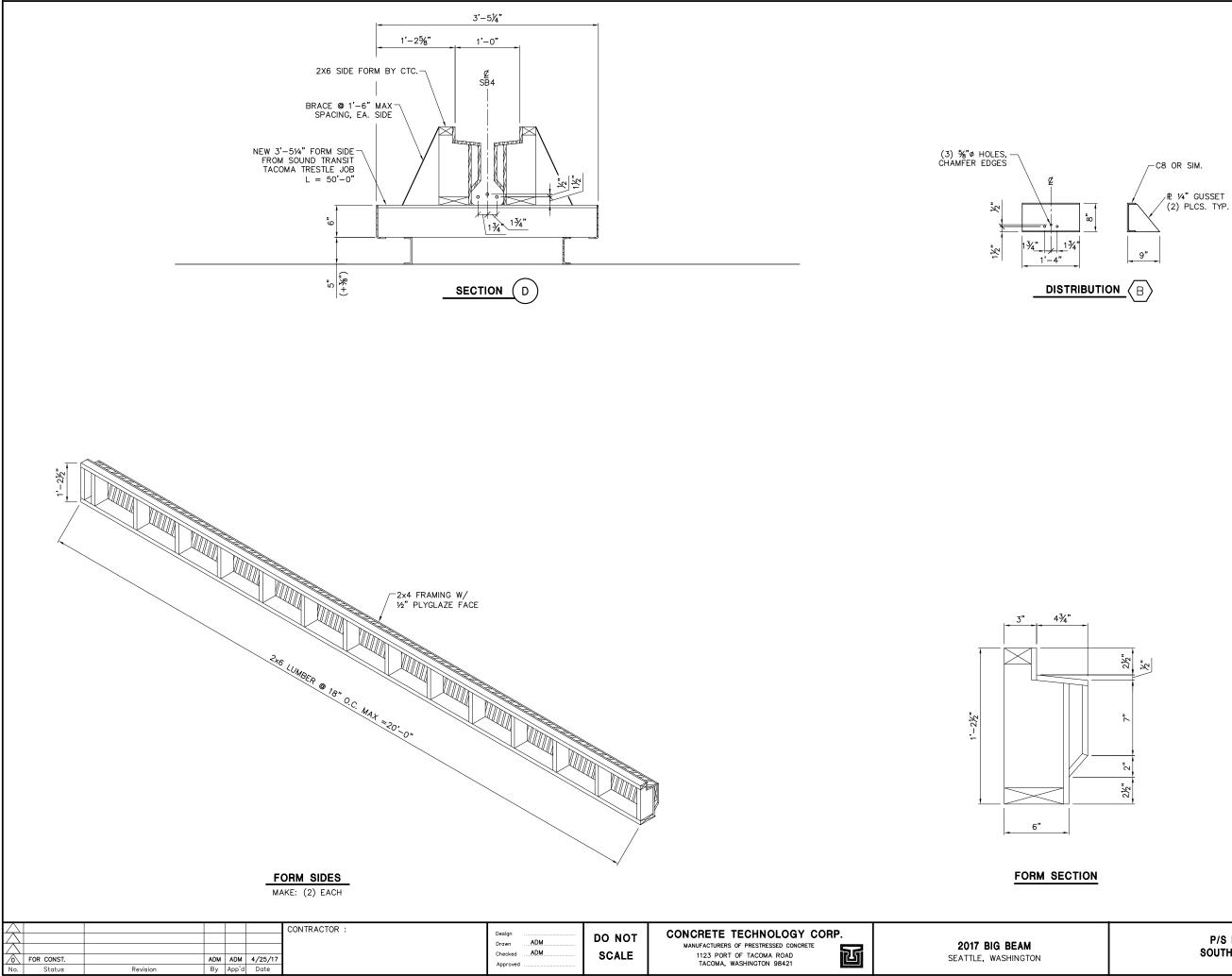
No. Status

Revision

Design Drawn Checked ADM MANUFACTURERS OF PRESTRESSED CONCRETE 1123 PORT OF TACOMA ROAD TACOMA, WASHINGTON 98421 ADM ADM 4/26/17 Ŀ "BIG BEAM" ADM SCALE 
 ADM
 ADM
 4/21/17

 By
 App'd
 Date
 Approved LACEY, WASHINGTON

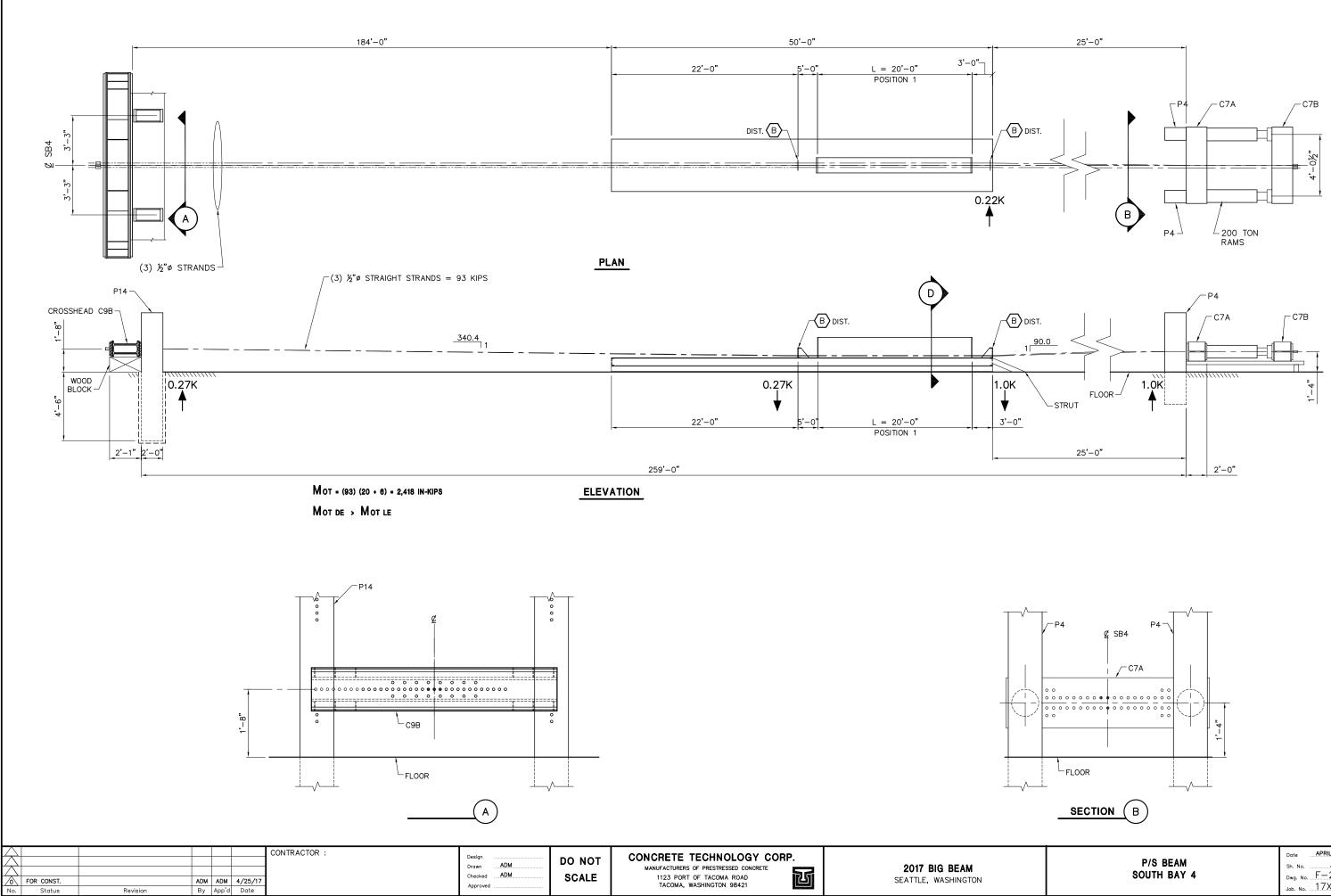
	REINFORCEMENT SCHEDULE										
GRADE	BAR MARK NUMBER	QUANTITY	STRAIGHT	NO. BENDS	CUTTING LENGTH	REBAR TYPE	BAR MAR	SIZ K N	E UMBER	<u>PIN</u> #3	$\frac{RUP/TIE}{\phi (UNO)} = 1\frac{1}{2}^{n}$ $= 2^{n}$
60	301	28		2	1'-6"					#5	= 21/2"
H	302 401	2	X	3	3'-5½" 20'-4"	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~					= 41/2" = 51/4"
H		2	ŕ	$\vdash$	20-4	Ś Ś	ŧ				= 594 $= 6^{"}$
Ħ	$\sim$								L		
$\vdash$	$\leq$					-1%"					
H	$\succ \prec$	]	┝								
H	$\succ \prec$	1	┢	┢							
H	$\geq$		t								
$\square$	$\square$					27⁄8"					
$\vdash$	$ \rightarrow $					(301)					
$\mathbb{H}$	$\succ \prec$		┢								
H	$\geq$										
$\square$	$\sim$										
$\mathbb{H}$	$\succ \prec$		┝	$\vdash$			ł			4"ø PIN	
H	$\geq$		┢								
t	$\sim$					- 434"					
Γ							1		Jl		11⁄2"ø PIN
1							<u>!</u>			_	11.1
								2"		2"	
								(	302	)	
									$\sim$		
1											
1											
	GENER	AL NO	TE	S							
-	ONCRETE:					7,000 P.S.I.					
ľ						. <u>13,000</u> P.S.I.					
<u> </u>	RESTRESS				, 7 WIRE U TION STRAN						
1		2011 NE				-					
1	STRAIGH	T: (3) STF	1AS	ND:	S JACKED T	0 93 KIPS (31.0 K/STR.)					
FI	NISHES:	TOP	S	TEF	L TROWEL						
"		SIDES	FC	DRM	I FINISH						
1		SOFFIT ENDS									
					PING BUNK						
_					FROM EAC						
<u>s</u>	HIPPING B	UNKING: 1	-	U″	FROM EAC	I LNU					
F	INSER	Г & AS	SE	EN	IBLY SC	HEDULE					
⊢					-						
$\vdash$	+						+	$\vdash$			
$\vdash$	$\rightarrow +$						-	$\vdash$			
⊢	$\searrow$						-				
L											
L											
L											
F							1	Ħ	PER		TOTAL
N	IARK				DESCR	PTION	PRIME	GALV.	PIECE	E	
⊢		6 . 11		_		21-11					TIES
	for which	e of these ch they we	pl ere	an pr	s and spec repared. A	fications shall be restricted to ny reproduction or distribution is	ine sex	orig	inal pu ssly lim	irpos ited	to
	such us part, w	se. Any o ithout the	th wr	er itt	reproduction en consent	ny reproduction or distribution is n, reuse or disclosure by any m of CTC is prohibited. These dr	etha awir	od, ngs	in whol and	e or	in
⊢	specific	ations con	τα	n	proprietary	information and title remains in	UI	U.			
L		PRO	DI	JC	TION D				APRIL 20		
B	EAMO <sup>.</sup>	THR				Sh	. No.	 , F	of 3M—2	· · · · ·	
<b> </b> №	IARK:_	BM2		Q	<b>TY.</b> <u>1</u>	_ WT. <u>1.55</u> KIPS [ 🦷	9. No.	1	17X08	ЗA	· · · · · · · · ·



Approved

Revision

P/S BEAM South Bay 4	Date         APRIL, 2017           Sh. No.         of           Dwg. No.         F-2.2           Job. No.         17X08
-------------------------	---



Date	APRIL, 2017
Sh. No.	of
Dwg. No	F-2.1 <u>/</u> U
Job. No.	17X08 A

# Appendix B

Weight and Cost

### **Concrete Properties:**

$$Volume = (Area)(Length) = \left(\frac{74.89 in^2}{144}\right) 20ft = 10.40 ft^3 = 0.352 yd^3$$
$$Weight_{Concrete} = (Volume)(\gamma_c) = (10.40 ft^3)(152.5 pcf) = 1586 lbs$$
$$Cost_{Concrete} = \left(\frac{\$120}{yd^3}\right)(0.352 yd^3) = \$46.23$$

### <u>Reinforcing Steel:</u>

#3 Bars

$$28 @ 1' - 6" = 42' - 0"$$

$$2 @ 3' - 5\frac{1}{2}" = 6' - 11"$$

$$Total \ Linear \ Feet = 48.92'$$

$$Weight_{\#3} = \left(0.376\frac{lb}{ft}\right)(48.92) = 18.39 \ lbs$$

#4 Bars

$$2 @ 20' - 0" = 40' - 0"$$

Total Linear Feet = 40.00'

$$Weight_{\#4} = \left(0.668 \frac{lb}{ft}\right)(40.00) = 26.72 \ lbs$$

$$Cost_{A615 Bar} = \left(\frac{\$0.45}{lb}\right)(18.39 \ lbs + 26.72 \ lbs) = \$20.30$$

Prestress Strand:

$$\frac{1}{2}$$
" Diameter:  
3 @ 20' = 60' - 0"  
 $Cost_{Strand} = \left(\frac{\$0.30}{ft}\right)(60') = \$18.00$ 

<u>Forming</u>

Sides = 39.7" \* 20' = 66.17 
$$ft^2$$
  
Ends = 2 @  $\frac{74.89 in^2}{144}$  = 1.04  $ft^2$   
Cost<sub>Forming</sub> =  $\left(\frac{\$1.25}{ft^2}\right)$  (67.21  $ft^2$ ) = \$84.01

### <u>Total Beam Weight</u>

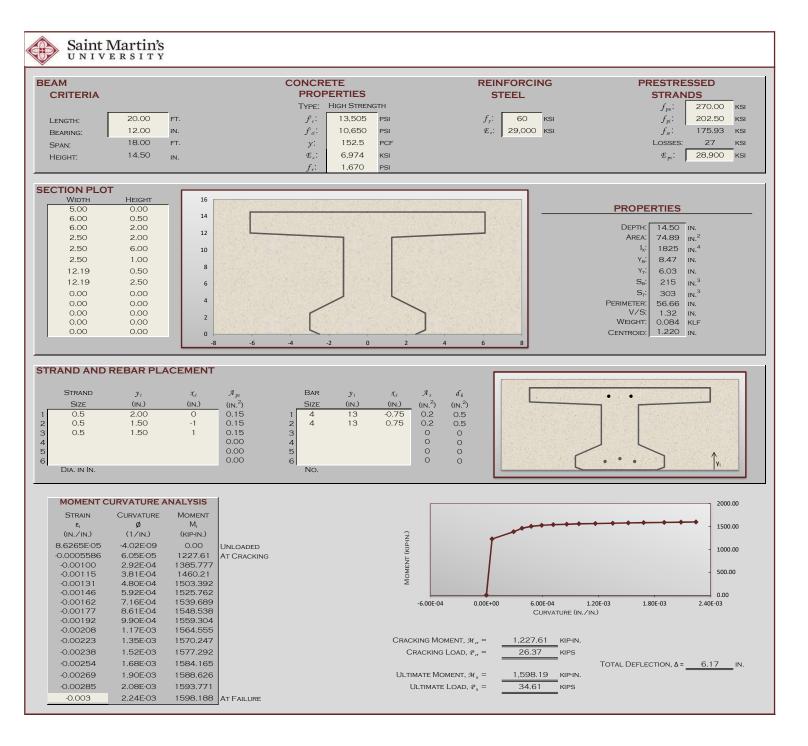
 $Weight_{Total} = W_{Concrete} + W_{#3} + W_{#4} = 1631 \ lbs = 82 \ plf$ 

### <u>Total Beam Cost</u>

 $Cost_{Concrete} = C_{Concrete} + C_{Bar} + C_{Strand} + C_{Forming} = $168.54$ 

# Appendix C

## Structural Design and Analysis Calculations



### SECTION PROPERTIES CALCULATOR

	AREA	YB	A <sub>YB</sub>	A(cg-y <sub>p</sub> ) <sup>2</sup>	I	I+A(Y <sub>BC</sub> ·Y <sub>B</sub> ) <sup>2</sup>	
	IN. <sup>2</sup>	IN.	IN. <sup>3</sup>	IN. <sup>4</sup>	IN. <sup>4</sup>	IN. <sup>4</sup>	Area Formulas:
SHAPE 1	2.75	0.26	0.71	185.49	0.06	185.54	RECTANGLE = $BH$
SHAPE 2	12.00	1.50	18.00	583.03	4.00	587.03	1
SHAPE 3	8.50	3.36	28.58	221.74	2.67	224.42	$T_{RAPEZOID} = \frac{1}{2}H(B_1 + B_2)$
SHAPE 4	15.00	7.50	112.50	14.12	45.00	59.12	2
SHAPE 5	2.50	11.00	27.50	16.00	0.21	16.21	Y <sub>B</sub> FORMULAS
SHAPE 6	3.67	11.80	43.35	40.83	0.07	40.90	RECTANGLE = 0.5 <sup>*</sup> H
SHAPE 7	30.47	13.25	403.71	696.06	15.87	711.93	$(H = R_{-} + 2R_{-})$
SHAPE 8			0.00	0.00		0.00	*TRAPEZOID = $H - \left(\frac{H}{3} * \frac{B_2 + 2B_1}{B_2 + B_1}\right)$
SHAPE 9			0.00	0.00		0.00	· · · · · · · · · · · · · · · · · · ·
SHAPE 10			0.00	0.00		0.00	MOMENT OF INERTIA FORMULAS:
Σ	74.89	_	634.35			1825.15	1
							RECTANGLE = $\frac{1}{12}BH^3$
	-	-					
A=	74.89						$T_{RAPEZOID} = \frac{H^3(B_2^2 + 4B_1B_2 + B_1^2)}{36(B_1 + B_2)}$
[=	1825.15						55(51 + 52)
CG =	1.220						
Y <sub>B</sub> =	8.47	IN.					
Y <sub>T</sub> =	6.03						
S <sub>n</sub> =	215.47	IN. <sup>3</sup>					
5,=	302.69	IN. <sup>3</sup>					
							$*B_1$ IS THE BOTTOM WIDTH
							FOR THE CG VALUE,
							+ INDICATES IT IS ABOVE D/2
PERIMETER CONTRIBUTI							- INDICATES IT IS BELOW D/2
SHAPE 1	6.41	IN.			TRAPEZOID		
SHAPE 2	4.00	IN.			RECTANGL		
SHAPE 3	5.32	IN.			TRAPEZOIE		
SHAPE 4	12.00	IN.			RECTANGL		
SHAPE 5	2.00	IN.			RECTANGL TRAPEZOIE		
SHAPE 6 SHAPE 7	9.74 17.19	IN. IN.			RECTANGL		
SHAPE /	56.656	IIN.		SHAPE 7 .	RECTANGL	E.	

SHAPE 1	6.41	IN.	S
SHAPE 2	4.00	IN.	S
SHAPE 3	5.32	IN.	S
SHAPE 4	12.00	IN.	S
SHAPE 5	2.00	IN.	S
SHAPE 6	9.74	IN.	S
SHAPE 7	17.19	IN.	S
Σ	56.656		
		-	



Saint Martin's UNIVERSITY

### MOMENT & CURVATURE CALCULATIONS

ε <sub>c</sub>	-0.003	in./in.
NEUTRAL AXIS, C	1.338	IN.
Moment @ $\varepsilon_c$	1,598	KIP-IN.
CURVATURE @ $\varepsilon_c$	0.002	1/in.

 $\Sigma$  Moments = 1,868 kip-in.

### 2 1/in.

y 14.5 13.16  $\varepsilon_c$  -0.003 0.00

### CROSS-SECTIONAL CONCRETE STRESS

SLICE NO. n <sub>i</sub>	НЕІGНТ <i>ћ</i> <sub>i</sub> (IN.)	WIDTH <i>δ<sub>i</sub></i> (IN.)	Dертн <i>у</i> і (IN.)	Strain $\varepsilon_i$ (in./in.)	STRESS $\sigma_i$ (PSI)	Force $F_i$ (LBS.)	Moment <i>M<sub>i</sub></i> (kip-in.)
1	0.290	12.19	14.36	-0.002675	13968	49,369	709
2	0.290	12.19	14.07	-0.002024	11412	40,336	567
З	0.290	12.19	13.78	-0.001374	7760.2	27,427	378
4	0.290	12.19	13.49	-0.000723	4086.2	14,442	195
5	0.290	12.19	13.20	-0.000073	412.18	1,457	19
6	0.290	12.19	12.91	0.000577	0	0	0
7	0.290	12.19	12.62	0.001228	0	0	0
8	0.290	12.19	12.33	0.001878	0	0	0
9	0.290	12.19	12.04	0.002529	0	0	0
10	0.290	7.247	11.75	0.003179	0	0	0
11	0.290	2.5	11.46	0.003830	0	0	0
12	0.290	2.5	11.17	0.004480	0	0	0
13	0.290	2.5	10.88	0.005131	0	0	0
14	0.290	2.5	10.59	0.005781	0	0	0
15	0.290	2.5	10.30	0.006432	0	0	0
16 17	0.290	2.5 2.5	10.01	0.007082	0	0	0
17	0.290 0.290	2.5 2.5	9.72 9.43	0.007732	0	0	0
19	0.290	2.5	9.43 9.14	0.009033	0	0	0
20	0.290	2.5	8.85	0.009684	0	0	0
21	0.290	2.5	8.56	0.010334	0	0	0
22	0.290	2.5	8.27	0.010985	Õ	Õ	Õ
23	0.290	2.5	7.98	0.011635	0	0	0
24	0.290	2.5	7.69	0.012286	0	0	0
25	0.290	2.5	7.40	0.012936	0	0	0
26	0.290	2.5	7.11	0.013586	0	0	0
27	0.290	2.5	6.82	0.014237	0	0	0
28	0.290	2.5	6.53	0.014887	0	0	0
29	0.290	2.5	6.24	0.015538	0	0	0
30	0.290	2.5	5.95	0.016188	0	0	0
31	0.290	2.5	5.66	0.016839	0	0	0
32	0.290	2.5	5.37	0.017489	0	0	0
33	0.290	2.5	5.08	0.018140	0	0	0
34	0.290	2.5	4.79	0.018790	0	0	0
35	0.290	2.509	4.50	0.019441	0	0	0
36	0.290	3.016	4.21	0.020091	0	0	0 0
37 38	0.290 0.290	3.524	3.92 3.63	0.020741	0	0	0
39	0.290	4.031 4.539	3.34	0.021392 0.022042	0	0	0
40	0.290	4.339 5.046	3.05	0.022693	0	0	0
41	0.290	5.554	2.76	0.023343	0	0	0
42	0.290	6	2.47	0.023994	Õ	0	0
43	0.290	6	2.18	0.024644	Õ	0	0
44	0.290	6	1.89	0.025295	0	0	0
45	0.290	6	1.60	0.025945	0	0	0
46	0.290	6	1.31	0.026596	0	0	0
47	0.290	6	1.02	0.027246	0	0	0
48	0.290	6	0.73	0.027896	0	0	0
49	0.290	5.87	0.44	0.028547	0	0	Ο
50	0.290	5.29	0.15	0.029197	0	0	0
ΣF	ORCES =	133	KIPS				
5 140	A AFAITO -	1000	LUD IN				

Concrete Forces =	-133.03	KIPS	
REBAR FORCES =	4.23	KIPS	
STRAND FORCES =	128.80	KIPS	
FOUILIBRIUM =	0 00	KIPS	

SECTI	ON PLC	т	
	Width	Height	
	5.00	0.00	
	6.00	0.50	
	6.00	2.50	
	2.50	4.50	
	2.50	10.50	
	2.50	11.50	
	12.19	12.00	
	12.19	14.50	
	0.00	14.50	
	#N/A	#N/A	



### Saint Martin's UNIVERSITY

STRAND	STRESS	SES					
Strand Dia. Ø <sub>ps</sub> (in.)	Dертн У <sub>і</sub> (IN.)	Conc. Strain <sub>ε,</sub> (in./in.)	TOT. STRAIN $\varepsilon_c + \varepsilon_{pe} + \varepsilon_{ce}$ (IN./IN.)	STRAND STRESS $\sigma_{ps}$ (KSI)	Strand Area A <sub>ps</sub> (in <sup>2</sup> )	STRAND FORCE ${\mathscr F}_i$ (KIPS)	Strand Moment M <sub>ps</sub> (kip-in)
0.5	2.00	0.0250	0.031551	280.5	0.153	42.92	85.83
0.5	1.50	0.0262	0.032672	280.7	0.153	42.94	64.42
0.5	1.50	0.0262	0.032672	280.7	0.153	42.94	64.42
0	0.00	0.00	0.00	0.00	0.00	0.00	0.00
0	0.00	0.00	0.00	0.00	0.00	0.00	0.00
0	0.00	0.00	0.00	0.00	0.00	0.00	0.00
				Σ =	0.459	IN. <sup>2</sup>	

**STRAND STRAIN** 

ε <sub>pe</sub> =	0.00609	IN./IN.
P <sub>e</sub> =	80.8	KIPS
M <sub>Beam</sub> =	3,410.21	FT-LBS
e =	6.80	IN
$\epsilon_{ce Top Fiber} =$	8.627E-05	in./in.
ε <sub>ce</sub> =	-0.000426	in./in.
$\epsilon_{ce Bottom Fiber} =$	-0.000493	in./in.
$\epsilon_{\rm pe}$ + $\epsilon_{\rm ce}$ =	0.00651	in./in.

#### POWER FORMULA

A = 156.96 кірз B = 28,743.04 кірз C = 104.31 D = 11.92

#### **REBAR STRESSES**

ΣFORCES = 128.80 KIPS ΣMOMENTS = 214.67 KIP-IN.

Bar Size	Dертн <i>у<sub>і</sub></i> (in.)	Strain €₃ (in.∕in.)	STRESS $\sigma_s$ (KSI)	CONC. STRESS <sub>o</sub> , (ksi)	Effective (ksi)	Steel Area A <sub>s</sub> (in <sup>2</sup> )	Steel Force $F_s$ (kips)	Rebar Moment M <sub>RS</sub> (kip-in)
4.0	13	0.0004	10.56759	0	10.5676	0.2	2.114	27.48
4.0	13	0.0004	10.56759	0	10.5676	0.2	2.114	27.48
0.0	0.0	0.0	0.0	0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0	0.0	0.0	0.0	0.0
					Σ =	0.400	IN. <sup>2</sup>	
ΣFORCES = ΣMOMENTS =	4.22704 54.9515							

## Saint Martin's

### DEFLECTION CALCULATIONS

P = W <sub>DL</sub> =	34.61 0.00702	KIPS KIPS/IN.				P = W <sub>DL</sub> =	0.00 0.00702	KIPS KIPS/IN.						
L =	216	IN.				L =	216	IN.						
OADED	)				1	UNLOAD	DED					MOMENT C	JRVATURI	E ANALYSIS
%	LENGTH	Moment	CURVATURE	Ø*dx*x,	Ø*dx*x,	%	LENGTH	MOMENT	CURVATURE	Ø*dx*x,	Ø*dx*x,	CURVATURE	MOMENT	
SPAN	Xi	$\mathcal{M}_i$	Ø	(IN.)	(IN.)	SPAN	Xi	$\mathcal{M}_i$	ø	(IN.)	(IN.)	Ø	$\mathcal{M}_i$	
0.00	(IN.)	(KIP-IN.)	(1 ∕ IN.)			0.00	(IN.)	(KIP-IN.)	(1 ∠ IN.)			(1/IN.)	(KIP-IN.)	
0.00 1.67	0.00 3.60	0.00 65.0	0.00	0.00 6.90E-06	0.00000	0.00	0.00 3.60	0.00 2.68	0.00	0.00 2.77E-07	0.00000	-4.02E-09 6.05E-05	0.00 1227.61	
3.33	7.20	129.9	3.20E-06 6.39E-06	5.52E-05	0.00007	3.33	7.20	5.27	1.28E-07 2.56E-07	2.21E-06	0.00000	2.92E-04	1385.78	AT CRACKIN
5.00	10.80	194.6	9.58E-06	1.45E-04	0.00017	5.00	10.80	7.78	3.79E-07	5.73E-06	0.00001	3.81E-04	1460.21	
6.67	14.40	259.3	1.28E-05	2.76E-04	0.00030	6.67	14.40	10.19	4.98E-07	1.07E-05	0.00001	4.80E-04	1503.39	
8.33	18.00	324.0	1.60E-05	4.48E-04	0.00048	8.33	18.00	12.50	6.12E-07	1.72E-05	0.00002	5.92E-04	1525.76	
10.00	21.60	388.5	1.91E-05	6.61E-04	0.00070	10.00	21.60	14.73	7.22E-07	2.49E-05	0.00003	7.16E-04	1539.69	
11.67	25.20	452.9	2.23E-05	9.15E-04	0.00096	11.67	25.20	16.87	8.27E-07	3.39E-05	0.00004	8.61E-04	1548.54	
13.33	28.80	517.2	2.55E-05	1.21E-03	0.00127	13.33	28.80	18.92	9.28E-07	4.41E-05	0.00005	9.90E-04	1559.30	
15.00	32.40	581.5	2.86E-05	1.55E-03	0.00161	15.00	32.40	20.87	1.02E-06	5.53E-05	0.00006	1.17E-03	1564.56	
16.67	36.00	645.6	3.18E-05	1.92E-03	0.00199	16.67	36.00	22.73	1.12E-06	6.75E-05	0.00007	1.35E-03	1570.25	
18.33	39.60	709.7	3.49E-05	2.34E-03	0.00242	18.33	39.60	24.51	1.20E-06	8.06E-05	0.00008	1.52E-03	1577.29	
20.00	43.20	773.7	3.81E-05	2.80E-03	0.00288	20.00	43.20	26.19	1.29E-06	9.44E-05	0.00010	1.68E-03	1584.16	
21.67	46.80	837.6	4.12E-05	3.30E-03	0.00339	21.67	46.80	27.78	1.36E-06	1.09E-04	0.00011	1.90E-03	1588.63	
23.33	50.40	901.4	4.44E-05	3.84E-03	0.00393	23.33	50.40	29.28	1.44E-06	1.24E-04	0.00013	2.08E-03	1593.77	
25.00	54.00	965.1	4.75E-05	4.41E-03	0.00452	25.00	54.00	30.69	1.51E-06	1.40E-04	0.00014	2.24E-03	1598.19	AT FAILURE
26.67	57.60	1,028.7	5.07E-05	5.03E-03	0.00514	26.67	57.60	32.01	1.57E-06	1.56E-04	0.00016			
28.33	61.20	1,092.2	5.38E-05	5.69E-03	0.00581	28.33	61.20	33.24	1.63E-06	1.73E-04	0.00018			
30.00	64.80	1,155.6	5.69E-05	6.39E-03	0.00651	30.00	64.80	34.37	1.69E-06	1.90E-04	0.00019	ΤΟΤΑ	L DEFLE	CTION
31.67	68.40	1,218.9	6.00E-05	7.13E-03	0.00726	31.67	68.40	35.42	1.74E-06	2.07E-04	0.00021			
33.33	72.00	1,282.2	1.40E-04	1.76E-02	0.01787	33.33	72.00	36.38	1.79E-06	2.24E-04	0.00023	$\Delta_T =$	6.17	IN.
35.00	75.60	1,345.3	2.33E-04	3.06E-02	0.03114	35.00	75.60	37.24	1.83E-06	2.41E-04	0.00025	-		=
36.67	79.20	1,408.4	3.19E-04	4.41E-02	0.04475	36.67	79.20	38.01	1.87E-06	2.58E-04	0.00026	$\Lambda_{-} \dots = \Lambda_{-}$	$\Delta_{\text{self Wei}} + \Delta_{\text{Self Wei}}$	+ Λ <sub>α</sub>
38.33	82.80	1,471.4	4.06E-04	5.88E-02	0.05968	38.33	82.80	38.69	1.90E-06	2.75E-04	0.00028	-Total -L	bad · —Self Wei	gnt · —Camper
40.00	86.40	1,534.3	6.68E-04	1.01E-01	0.10243	40.00	86.40	39.29	1.93E-06	2.92E-04	0.00030			
41.67	90.00	1,597.1	2.20E-03	3.47E-01	0.35194	41.67	90.00	39.79	1.96E-06	3.08E-04	0.00031			
43.33	93.60	1.597.5	2.22E-03	3.64E-01	0.36867	43.33	93.60	40.20	1.98E-06	3.24E-04	0.00033			
45.00	97.20	1,597.8	2.23E-03	3.80E-01	0.38502	45.00	97.20	40.51	1.99E-06	3.40E-04	0.00034			
46.67	100.80	1,598.0	2.24E-03	3.96E-01	0.40093	46.67	100.80	40.74	2.00E-06	3.55E-04	0.00036			
48.33	104.40	1,598.1	2.24E-03	4.12E-01	0.41634	48.33	104.40	40.88	2.01E-06	3.69E-04	0.00037			
50.00	108.00	1,598.2	2.24E-03	4.26E-01	0.43118	50.00	108.00	40.92	2.01E-06	3.82E-04	0.00039			
			Σ =	2.62540	2.65939	_			Σ =	0.00490	0.00499			
		1	$\Delta_{LOAD} =$	5.28480	IN.				$\Delta_{SELF WEIGHT} =$	0.00990	IN.	$\Delta_{CAMBER} =$	0.875	IN.
			LOND						SELF WEIGHT					Y MEASURED
00	LOADE	d Curva	TURE VRS.	LENGTH			Unloai	ded Cuf	RVATURE VR	S. LENGT	H I			
00				•	• • • • •	0.00								
00						0.00								
00						0.00				•••				
00						0.00			******					
				•		0.00								
00						-0.00								
00 • • •						-12 0.00		28	48	68 88	108			
00 0	20	40	60	80	100	0.00								
		SPAN I	Length (in.)					SP/	N LENGTH (IN.)					



## Saint Martin's

### SHEAR CALCULATIONS

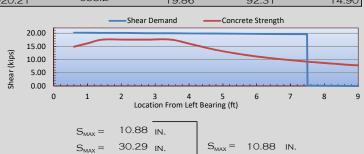
[	$V_u =$	39,000	LBS.	6 <sub>w</sub> =	2.50	IN.	P <sub>u</sub> =	39.00 KIPS	$f_{pe} =$	3.63	KSI
	$\mathcal{V}_c =$	7457	LBS.	<i>d</i> =	12.83	IN.	W <sub>DL</sub> =	0.084 KIPS/IN.	$V_p =$	0.00	KIPS
	$\varphi =$	0.75									

	х		APPLIED SHEAR (KIF	rs)	$\mathcal{M}_{cr}$	She	ear Strength (kip	S)
	(FT.)	$\mathcal{V}_{DL}$	$\mathcal{V}_{\scriptscriptstyle \mathcal{LL}}$	$\mathcal{V}_{\mathcal{V}}$	(KIP-IN.)	$\mathcal{V}_{cw}$	$\mathcal{V}_{\dot{a}}$	$\mathcal{GV}_c$
h/2	0.60	0.71	19.50	20.21	658.2	19.86	93.72	14.90
	1.02	0.67	19.50	20.17	789	21.65	67.12	16.23
ĺt	1.44	0.64	19.50	20.14	919.9	23.43	55.98	17.57
	2.12	0.58	19.50	20.08	915	23.43	38.85	17.57
	2.79	0.52	19.50	20.02	910.6	23.43	29.97	17.57
	3.46	0.47	19.50	19.97	906.6	23.43	24.53	17.57
	4.13	0.41	19.50	19.91	903	23.43	20.85	15.64
ld	4.81	0.35	19.50	19.85	900	23.43	18.20	13.65
	5.64	0.28	19.50	19.78	896.8	23.43	15.76	11.82
	6.48	0.21	19.50	19.71	894.3	23.43	13.94	10.46
	7.49	0.13	19.50	19.63	892.2	23.43	12.29	9.218
	7.50	0.13	0.00	0.13	892.2	23.43	12.28	9.208
	8.34	0.06	0.00	0.06	891.3	23.43	11.20	8.4
Midspan	9.00	0.00	0.00	0.00	891.1	23.43	10.49	7.866
	9.84	-0.07	0.00	-0.07	891.3	23.43	11.27	8.451
	10.68	-0.14	0.00	-0.14	892.2	23.43	12.25	9.188
	11.52	-0.21	-19.50	-19.71	894.3	23.43	13.52	10.14
	12.36	-0.28	-19.50	-19.78	896.8	23.43	15.19	11.4
ld	13.19	-0.35	-19.50	-19.85	900	23.43	17.49	13.12
	13.87	-0.41	-19.50	-19.91	903	23.43	20.03	15.03
	14.54	-0.47	-19.50	-19.97	906.6	23.43	23.60	17.57
	15.21	-0.52	-19.50	-20.02	910.6	23.43	28.93	17.57
	15.88	-0.58	-19.50	-20.08	915	23.43	37.69	17.57
ĺt	16.56	-0.64	-19.50	-20.14	919.9	23.43	54.71	17.57
	16.98	-0.67	-19.50	-20.17	789	21.65	65.78	16.23
h/2	17.40	-0.71	-19.50	-20.21	658.2	19.86	92.31	14.90

#### STIRRUP DESIGN

 $\mathcal{GV}_s \operatorname{REQ.} \geq \mathcal{V}_u - \mathcal{GV}_c$ 

LOCATION	фVs	Req.
Ат <i>ћ</i> /2	5.31	KIPS
AT <i>Lt</i>	2.57	KIPS
AT Ld	6.21	KIPS
AT MIDSPAN	-7.87	KIPS



CONTROLS

S<sub>MAX</sub> = 24.13 IN.

### SPACING REQUIREMENTS

LOCATION	$S_{REQ.}$		Spacing		фVs		SPACING USAGE
AT <i>h</i> /2	11.96	IN.	10.00 in	Ν.	6.353	KIPS	Use 10 in. Near Ends
AT <i>Lt</i>	24.76	IN.	10.00 in	Ν.	6.353	KIPS	USE 10 IN. TO TRANSITION
AT Ld	10.24	IN.	10.00 in	Ν.	6.353	KIPS	Use 10 in. Near Ld
AT MIDSPAN	-8.08	IN.	10.00 in	Ν.	6.353	KIPS	Use 10 in. Middle



## Saint Martin's UNIVERSITY

PA	DA	B A D	 DC
ГА	RA		<b>R</b> 3

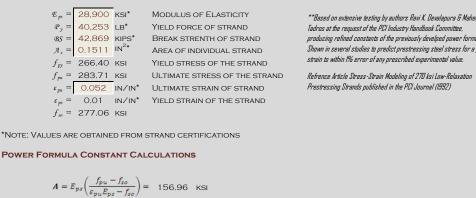
PARAME	TERS		
	Area =	74.89063	$IN.^2$ Age @ Release = 92 Hrs No. Strands = 3 0.5" Dia.
		1,825.15	
H	EIGHT =	14.50	IN. AGING COEFF., $\chi = 0.7$ $f_{pu} = 270$ KSI
	w =	84.2	PLF HUMIDITY = 75 % $\mathcal{E}_{ps}$ = 28,900 KSI
	V/S=	1.3219	IN. $K1 = 1.0$ $y_c = 1.67$ IN
LE	NGTH =	20.00	FT. K2 = 1.0 f <sub>ру</sub> = 243.0 кsi
			$f_{jj} = 202.5$ KSI
Prestri	ESS LOS	SES	
	$K_f =$	0.43	Concrete Strenght Factor $\chi_{M} = 0.57$ Time Development Factor
ш	$\kappa_s =$	1.28	Size Factor $\chi_{fa} = 0.85$ Loading Factor
KAC	$\mathcal{K}_{hs} =$	0.93	HUMIDITY FACTOR FOR SHRINKAGE $\chi_{hc} = 0.96$ HUMIDITY FACTOR FOR CREEP
SHRINKAGE	$K_{td} =$	0.60	HUMIDITY FACTOR FOR SHRINKAGE $\chi_{fe} = 0.96$ HUMIDITY FACTOR FOR CREEPTIME DEVELOPMENT FACTOR $\chi_{fe} = 0.43$ CONCRETE STRENGTH FACTOR $\chi_{fe} = 1.28$ SIZE FACTOR
R	$y_{sh} =$	0.31	
	$\varepsilon_{sh} =$	1.47E-04	Shrinkage strain at testing $y_{\sigma} = 0.26$
			$\psi_{\alpha} = 0.48$
TRANSFORMED SECTION @ TRANSFER	$w_c =$	153.5	PCF $A_{ti} = 76.48 \text{ IN.}^2$ $K_r = 0.942$
PEF 0	$\mathcal{E}_{ci} =$	6,477	KSI $\Upsilon_{bii} = 8.33$ IN. $\mathcal{K}_{nd} = 0.924$
SFG	$\mathcal{E}_{c} =$	7,294	KSI $I_{ii} = 1,897 \text{ IN.}^4$
TR/ SEC	$n_i =$	4.46	MODUALR RATIO $e_{pti} = 6.66$ IN.
Ë.	n =	3.96	MODULAR RATIO $e_p = 6.80$ IN.
			$\alpha = 2.90$
LR =	$\frac{f_{pj}}{45} \left(\frac{f_{pj}}{f_{py}}\right)$	$-0.55 \times 10^{-10}$	$\log\left(\frac{Age\ at\ Release\ (hours)\ +\ 1}{1}\right) = 2.51$ ksi Relaxation prior to transfer
			$f_{pi} = 199.99$ ksi Stress just before transfer
		$\Delta ES_n = \frac{P}{2}$	$\frac{i}{4} \alpha K_r n_i = 14.938$ ksi Elastic shortening (from prestress)
	4		$\frac{R}{p_p}K_r n_i = -0.066$ ksi Elastic shortening (from self weight)
			•
	ΔC	$R_{bd} = n_i f_{cir}$	$\psi_{cr}K_{rd} = 6.66$ ksi Creep
LR =	$\frac{f_{pj}}{45} \left( \frac{f_{pj}}{f_{py}} \right)$	$-0.55 \times 10^{-10}$	$\log\left(\frac{Age\ at\ 28\ Days\ (hours)+1}{Hours\ at\ Transfer\ +1}\right) = 1.1 \qquad \text{KSI}  \text{Relaxation losses after transfer}$
		TOTAL L	DSSES = 26.57 KSI



### Saint Martin's

#### STRESS-STRAIN MODELING OF 270 KSI LOW-RELAXATION PRESTRESSING STRANDS - POWER FORMULA

#### MATERIAL PARAMETERS



\*\*Based on extensive testing by authors Ravi K. Devalapura & Maher K. producing refined constants of the previously develped power formula. Shown in several studies to predict prestressing steel stress for a given

Prestressing Strands published in the PCI Journal (1992)

 $C = \frac{E_{ps}}{f_{so}} = 104.31$ 

 $B = E_{ps} - A = 28,743$  KSI

$$D = 11.92 \leftarrow$$

$$f_{ps} = \varepsilon_{ps} \left( A + \frac{B}{\left(1 + \left(C\varepsilon_{ps}\right)^{D}\right)^{\frac{1}{D}}} \right) = 266.40 \text{ kSI}$$

$$f_{py} = 266.40 \text{ kSI}$$

ITERATE VALUES OF "
$$D$$
 " UNTIL  $f_{ps} = f_{pp}$   
Once done hit the 'Run Analysis' button  
on the 'Beam Section' sheet.

Saint Martin's

 $\begin{aligned} &Top \ Flange \ Stress = f_t = \frac{P}{A} - \frac{Pe}{S_t} + \frac{M_{self-weigh}}{S_t} \\ &Bottom \ Flange \ Stress = f_b = \frac{P}{A} + \frac{Pe}{S_b} - \frac{M_{self-weigh}}{S_b} \end{aligned}$ 

 $P = \begin{bmatrix} 83.65 & \text{KIPS} \\ E = & 6.80 & \text{IN.} \\ A = & 74.89 & \text{IN}^2 \\ S_T = & 302.69 & \text{IN}^3 \\ S_B = & 215.47 & \text{IN}^3 \end{bmatrix}$ 

 $\label{eq:allowable} \begin{array}{l} \textit{Allowable Tension Stress} = 6 \sqrt{f_c'} @ \textit{Ends} \\ \textit{Allowable Tension Stress} = 3 \sqrt{f_c'} @ \textit{Otherwise} \\ \textit{Allowable Compression Stress} = 0.6 f_c' \end{array}$ 

SIGN CONVENTION : + = COMPRESSION -= TENSION

\*ASSUMING CONCRETE STRENGTH AT RELEASE IS 10650 PSI

TOP OF	BEAM									Воттом	OF BEAM					
% Span	Length <i>X<sub>i</sub></i> (in.)	Moment $\mathcal{M}_i$	$f_{SELF}$ Weight	f <sub>Prestress</sub> (KSI)	$f_{TOTAL}$ (KSI)	$f_{ALLOWABLE}$ (KSI)	Min. <i>fc</i> Required	AREA OF STEEL REQUIRED		% Span	Length <sub><i>X<sub>i</sub></i> (in.)</sub>	Moment $\mathcal{M}_i$	$f_{SELF}$ Weight	f <sub>Prestress</sub> (KSI)	$f_{TOTAL}$ (KSI)	Min <i>fc</i> Required
	(IN.)	(KIP-IN.)	(KSI)	(1(3))	(1(3))	(K3I)	(KSI)	(IN <sup>2</sup> )			(IN.)	(KIP-IN.)	(KSI)	(1(3))	(R31)	(KSI)
0.00	0.00	0.00	0.00	0.00	0.00	-0.62	0.00	0.00		0.00	0.00	0.00	0.00	0.00	0.00	0.00
1.16	2.78	-0.03	0.00	-0.08	-0.08	-0.62	0.20	0.04		1.29	2.78	-0.03	0.00	0.42	0.42	0.70
2.31	5.56	-O.11	0.00	-0.17	-0.17	-0.62	0.80	0.08		2.57	5.56	-O.11	0.00	0.84	0.84	1.39
3.47	8.33	-0.24	0.00	-0.25	-0.26	-0.62	1.81	0.13		3.86	8.33	-0.24	0.00	1.25	1.25	2.09
4.63	11.11	-0.43	0.00	-0.34	-0.34	-0.62	3.22	0.17		5.14	11.11	-0.43	0.00	1.67	1.67	2.79
5.79	13.89	0.91	0.00	-0.42	-0.42	-0.62	4.92	0.21		6.43	13.89	0.91	0.00	2.09	2.08	3.47
6.94	16.67	2.95	0.01	-0.51	-0.50	-0.62	6.92	0.25		7.72	16.67	2.95	-0.01	2.51	2.49	4.15
8.10	19.44	4.94	0.02	-0.59	-0.58	-0.62	9.26	0.28		9.00	19.44	4.94	-0.02	2.92	2.90	4.83
9.26	22.22	6.87	0.02	-0.68	-0.66	-0.62	11.94	0.32		10.29	22.22	6.87	-0.03	3.34	3.31	5.51
10.42	25.00	8.75	0.03	-0.76	-0.73	-0.62	14.98	0.36		11.57	25.00	8.75	-0.04	3.76	3.72	6.20
12.22	29.32	11.57	0.04	-0.76	-0.73	-0.31	58.41	0.35		13.57	29.32	11.57	-0.05	3.76	3.70	6.17
14.02	33.64	14.25	0.05	-0.76	-0.72	-0.31	56.99	0.34		15.57	33.64	14.25	-0.07	3.76	3.69	6.15
15.81	37.95	16.80	0.06	-0.76	-0.71	-0.31	55.66	0.34		17.57	37.95	16.80	-0.08	3.76	3.68	6.13
17.61	42.27	19.22	0.06	-0.76	-0.70	-0.31	54.41	0.33		19.57	42.27	19.22	-0.09	3.76	3.67	6.12
19.41	46.59	21.51	0.07	-0.76	-0.69	-0.31	53.24	0.32		21.57	46.59	21.51	-0.10	3.76	3.66	6.10
21.21	50.91	23.67	0.08	-0.76	-0.69	-0.31	52.15	0.32		23.57	50.91	23.67	-0.11	3.76	3.65	6.08
23.01	55.23	25.70	0.08	-0.76	-0.68	-0.31	51.13	0.31		25.57	55.23	25.70	-0.12	3.76	3.64	6.07
24.81	59.55	27.59	0.09	-0.76	-0.67	-0.31	50.19	0.31		27.57	59.55	27.59	-0.13	3.76	3.63	6.05
26.61	63.86	29.36	0.10	-0.76	-0.67	-0.31	49.32	0.30		29.57	63.86	29.36	-0.14	3.76	3.62	6.04
28.41	68.18	31.00	0.10	-0.76	-0.66	-0.31	48.53	0.30		31.57	68.18	31.00	-0.14	3.76	3.61	6.02
30.21	72.50	32.50	O.11	-0.76	-0.66	-0.31	47.80	0.30		33.56	72.50	32.50	-0.15	3.76	3.61	6.01
32.01	76.82	33.88	O.11	-0.76	-0.65	-0.31	47.14	0.29		35.56	76.82	33.88	-0.16	3.76	3.60	6.00
33.81	81.14	35.12	0.12	-0.76	-0.65	-0.31	46.55	0.29		37.56	81.14	35.12	-0.16	3.76	3.60	5.99
35.61	85.45	36.23	0.12	-0.76	-0.64	-0.31	46.02	0.29		39.56	85.45	36.23	-0.17	3.76	3.59	5.98
37.41	89.77	37.21	0.12	-0.76	-0.64	-0.31	45.56	0.29		41.56	89.77	37.21	-0.17	3.76	3.59	5.98
39.20	94.09	38.06	0.13	-0.76	-0.64	-0.31	45.16	0.28		43.56	94.09	38.06	-0.18	3.76	3.58	5.97
41.00	98.41	38.78	0.13	-0.76	-0.64	-0.31	44.82	0.28		45.56	98.41	38.78	-0.18	3.76	3.58	5.96
42.80	102.73	39.37	0.13	-0.76	-0.63	-0.31	44.55	0.28		47.56	102.73	39.37	-0.18	3.76	3.58	5.96
44.60	107.05	39.83	0.13	-0.76	-0.63	-0.31	44.34	0.28		49.56	107.05	39.83	-0.18	3.76	3.57	5.96
46.40	111.36	40.16	0.13	-0.76	-0.63	-0.31	44.18	0.28		51.56	111.36	40.16	-0.19	3.76	3.57	5.95
48.20	115.68	40.35	0.13	-0.76	-0.63	-0.31	44.09	0.28		53.56	115.68	40.35	-0.19	3.76	3.57	5.95
50.00	120.00	40.42	0.13	-0.76	-0.63	-0.31	44.06	0.28		55.56	120.00	40.42	-0.19	3.76	3.57	5.95
						REINFORCEMENT		0.36						ETE STRENGTH		6.20
					AREA OF F	REINFORCEMENT	PROVIDED =	0.40	• Oł	<			CONCE	ETE STRENGTH	PROVIDED =	10.65
													22.00			

# Appendix D

Concrete Mix Design

### BATCH REPORT by Batch Number

### Concrete Technology Corporation, Tacoma, WA

Cast Date:	5/5/2017	Mixer Number:	2	Station Number:	2	
DB ID#:	22378	Call Time:	2:14:33	3 PM		
Recipe Number:	140	Mix Start Time:	2:21:17	7 PM		
Recipe Name:	140	Complete Time:	2:23:33	3 PM		
Daily Count No.:	64	Discharge Time:	2:24:38	3 PM		
Batches this Pour:	1	W/C Target:	0.270			
Yards this Pour:	2.3	W/C Actual:	0.268			
Yards This Batch:	2.3	Water Temperature:	59.3	°F		
Job Number:	17X8O	Batched	in Auto:	Mixed i	n Auto: 🗹	Hot Mix Alarm: 🗌
Job Name:	BIG BEAM					

### AGGREGATES

Mark Number:

AG	GREGATES	SSD Target	SSD Actual	Dev.	Free Water	Total Moisture	Absorbed Moisture	Actual Wet Wt.
	Name	lbs.	lbs.	%	lbs.	%	%	lbs.
1	5/8"	2,270	2,274	0.18%	23	2.00	0.95	2,297
2	5/8"	2,270	2,265	-0.22%	23	2.00	0.95	2,288
3	Sand	1,393	1,373	-1.44%	64	6.50	1.85	1,437
4	Sand	1,396	1,393	-0.21%	78	7.47	1.85	1,471
5	#8 PEA GRAVEL	0	0	0.00%	0	0.00	0.00	0
6	#8 PEA GRAVEL	0	0	0.00%	0	0.00	0.00	0
	TOTAL	7,329	7,305		188			7,493

CE	MENTS				AD	MIXTURES				
	Name	Target lbs.	Actual lbs.	Dev. %		Name	Target oz.	Actual oz.	Dev. %	Water %
1	Silica Fume	0	0	0.00%	2.1	Daravair 1000	0.0	0.0	0.00%	0.0%
2	Fly Ash	0	0	0.00%	2.2	WDRA 64	69.0	69.0	0.00%	0.0%
3	TYPE III	0	0	0.00%	2.3	DCI	0.0	0.0	0.00%	0.0%
4	TYPE III	1,729	1,724	-0.29%	2.4	VMAR	0.0	0.0	0.00%	0.0%
	TOTAL	1,729	1,724	-0.29%	2.5	ADVA 575	156.0	156.0	0.00%	0.0%

										Max. Prot Target	be
WATER Total Metered Target	Adjusted Metered Target	Metered Actual	Dev. %	Probe Metered Actual	Manual Metered Actual	Total Metered Actual	Aggregate Moisture	Admixture Moisture	TOTAL Water Actual	0 Probe Readings	
55.9 gal.	33.2 gal.	33.1 gal.	-0.30%	0.0 gal.	0.0 gal.	33.0 gal.	22.6 gal.	0.0 gal.	55.6 gal.	0	at Final mix
465 lb.	277 lb.	276 lb.		0 lb.	0 lb.	275 lb.	188 lb.	0 lb.	463 lb.	0	at Discharge

operator

#### 

Page 1 of 1

# Appendix E

Strand Specifications



**Prestressed Concrete Strand Division** 

East: 710 Marshall Stuart Drive, Dickson, TN 37055 • 866-491-5020 West: 1412 El Pinal Drive, Stockton, CA 95205 • 866-246-3758

Thipon 2 No samples needed

### MILL CERTIFICATE OF INSPECTION

Order Number:	SLPC170283-1	Page No :	1 OF 1
B/L No:	SIPC171044	Issue Date :	04/25/2017
Commodity:	Steel Strand, Uncoated Seven Wire for	Prestressed Conc	rete
Size & Grade:	1/2" x 270 KSI		
Specification:	ASTM A416-Latest 1/2"-Low Relaxation		
Customer Name:	CONCRETE TECHNOLOGY CORPORATION		
Customer P.O.:	6-04151		
Destination:	CONTEC-WA		
State Job No:			

No	Pack #	Heat #	B.S.	Elong.	Υ.Ρ.	Area	E-Modulus	CURVE#
			Min:41,300 (LB)	3.5 (%)	37,170 (LB)	(IN2)	(MPSI)	
1	S129195-5	S0286957		4.6	39,844	0.1513	28.8	S129195
2	S529386-3	S0286956	43,291	5.0	40,411	0.1514	28.8	S529386
<b>*</b> 3	S529394-6	S0286409	42,869	5.2	40,253	0.1511	28.9	S529394
4	S529395-3	S0286957	42,732	4.9	40,282	0.1513	28.9	S529395
5	S529396-1	S0286957	42,829	5.4	40,274	0.1516	28.7	S529396
6	S529396-2	S0286957	42,829	5.4	40,274	0.1516	28.7	S529396
7	s529398-7	S0286959	42,833	5.5	40,094	0.1520	28.7	S529398

#### We hereby certify that:

\* We have accurately carried out the inspection of COMMODITY and met the requirements in accordance with the applicable SPECIFICATION, both listed above.

 $\star$  The material described above will bond to concrete of a normal strength and consistency in conformance with the prediction equations for transfer and development length given in the ACI/AASHTO specifications.

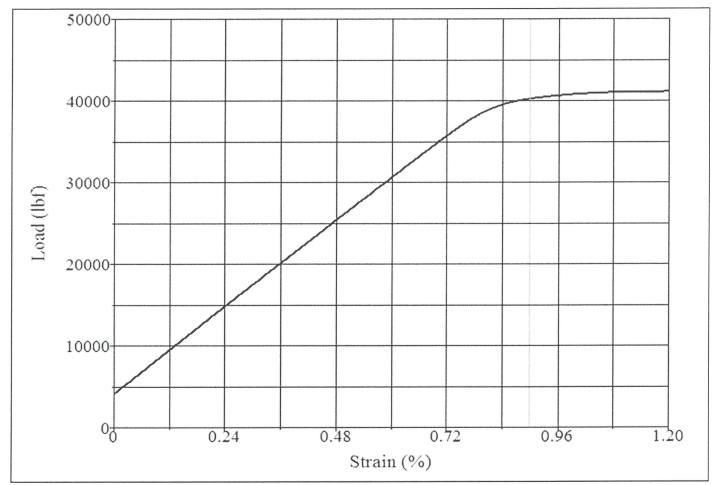
 $\ast$  The individual below has the authority to make this certificate legally binding for SWPC.

Received by, Net II. Milallon

Date: 4/25/17 CMO: NO PO: 6-04151 Job: Inventory Item: Strand 1/2" Commercial



**Prestressed Concrete Strand Division** East: 710 Marshall Stuart Drive, Dickson, TN 37055 • 866-491-5020 West: 1412 El Pinal Drive, Stockton, CA 95205 • 866-246-3758



\*Vertical Line is drawn at 1% Extension Under Load

### Curve # \$529394

Yield Point	40253	lbf
Area	0.1511	in²
Modulus	28.9	Msi

## CONTACT INFORMATION

JILL WALSH, PhD, PE FACULTY ADVISOR	CAMERON B. REECE PROJECT MANAGER/LEAD ENGINEER
Hal & Inge Marcus School of Engineering Saint Martin's University	6440 SE Lynch Rd. Shelton, WA 98584
5000 Abbey Way SE	Cameron.Reece@stmartin.edu
Lacey, WA. 98503	360.545.5336
JWalsh@stmartin.edu	
WILLIAM J. MILLER III	PAUL RUMBLES
ASSISTANT MANAGER	ASSISTANT ENGINEER
15351 SE Callie Ave	920 SW Ferry St
Yelm, WA 98597	Tumwater, WA 98512
William.Miller@stmartin.edu	Paul.Rumbles@stmartin.edu
253.973.4596	360.628.9023
JARAD ROSCHI	JOEL ROGERS
REPORT EDITOR	FABRICATOR
REPORT EDITOR	
5712 SE Acarro Ct	9241 NE Skokomish Way #2
	9241 NE Skokomish Way #2 Lacey, WA 98516
5712 SE Acarro Ct	-
5712 SE Acarro Ct Lacey, WA 98503	Lacey, WA 98516
5712 SE Acarro Ct Lacey, WA 98503 Jarad.Roschi@stmartin.edu	Lacey, WA 98516 JoelB.Rogers@stmartin.edu
5712 SE Acarro Ct Lacey, WA 98503 <u>Jarad.Roschi@stmartin.edu</u> 317.372.7938	Lacey, WA 98516 JoelB.Rogers@stmartin.edu 719.393.3793
5712 SE Acarro Ct Lacey, WA 98503 <u>Jarad.Roschi@stmartin.edu</u> 317.372.7938 CLARINDA MARION	Lacey, WA 98516 JoelB.Rogers@stmartin.edu 719.393.3793 DAVID ROWLAND
5712 SE Acarro Ct Lacey, WA 98503 <u>Jarad.Roschi@stmartin.edu</u> 317.372.7938 CLARINDA MARION LEAD VIDEOGRAPHER	Lacey, WA 98516 <u>JoelB.Rogers@stmartin.edu</u> 719.393.3793 DAVID ROWLAND LEAD DRAFTER
5712 SE Acarro Ct Lacey, WA 98503 <u>Jarad.Roschi@stmartin.edu</u> 317.372.7938 CLARINDA MARION LEAD VIDEOGRAPHER 1139 Bell Hill Place	Lacey, WA 98516 <u>JoelB.Rogers@stmartin.edu</u> 719.393.3793 DAVID ROWLAND LEAD DRAFTER 9541 NE 39 <sup>th</sup> Loop
5712 SE Acarro Ct Lacey, WA 98503 <u>Jarad.Roschi@stmartin.edu</u> 317.372.7938 CLARINDA MARION LEAD VIDEOGRAPHER 1139 Bell Hill Place Dupont, WA 98327	Lacey, WA 98516 JoelB.Rogers@stmartin.edu 719.393.3793 DAVID ROWLAND LEAD DRAFTER 9541 NE 39 <sup>th</sup> Loop Olympia, WA 98516